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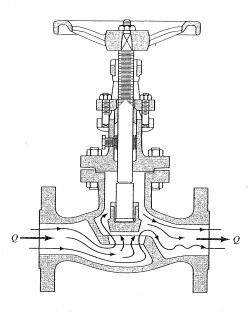


FIGURE 8.21 Flow through a valve.

or

$$h_{L\,\text{minor}} = K_L \frac{V^2}{2g} \tag{8.36}$$

The pressure drop across a component that has a loss coefficient of  $K_L = 1$  is equal to the dynamic pressure,  $\rho V^2/2$ .

The actual value of  $K_L$  is strongly dependent on the geometry of the component considered. It may also be dependent on the fluid properties. That is,

$$K_L = \phi(\text{geometry}, \text{Re})$$

where  $Re = \rho VD/\mu$  is the pipe Reynolds number. For many practical applications the Reynolds number is large enough so that the flow through the component is dominated by inertia effects, with viscous effects being of secondary importance. This is true because of the relatively large accelerations and decelerations experienced by the fluid as it flows along a rather curved, variable area (perhaps even torturous) path through the component (see Fig. 8.21). In a flow that is dominated by inertia effects rather than viscous effects, it is usually found that pressure drops and head losses correlate directly with the dynamic pressure. This is the reason why the friction factor for very large Reynolds number, fully developed pipe flow is independent of the Reynolds number. The same condition is found to be true for flow through pipe components. Thus, in most cases of practical interest the loss coefficients for components are a function of geometry only,  $K_L = \phi(\text{geometry})$ .

Minor losses are sometimes given in terms of an *equivalent length*,  $\ell_{eq}$ . In this terminology, the head loss through a component is given in terms of the equivalent length of pipe that would produce the same head loss as the component. That is,

$$h_{L\,\text{minor}} = K_L \frac{V^2}{2g} = f \frac{\ell_{\text{eq}}}{D} \frac{V^2}{2g}$$

or

For most flows the

loss coefficient is

independent of the

Reynolds number.

$$\ell_{\rm eq} = \frac{K_L D}{f}$$

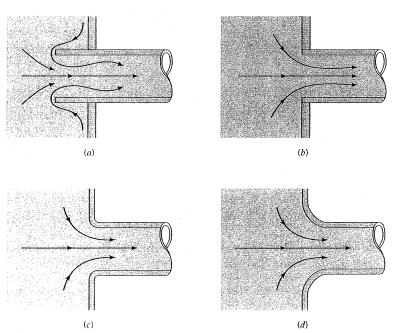
where D and f are based on the pipe containing the component. The head loss of the pipe system is the same as that produced in a straight pipe whose length is equal to the pipes of the original system plus the sum of the additional equivalent lengths of all of the components of the system. Most pipe flow analyses, including those in this book, use the loss coefficient method rather than the equivalent length method to determine the minor losses.

Many pipe systems contain various transition sections in which the pipe diameter changes from one size to another. Such changes may occur abruptly or rather smoothly through some type of area change section. Any change in flow area contributes losses that are not accounted for in the fully developed head loss calculation (the friction factor). The extreme cases involve flow into a pipe from a reservoir (an entrance) or out of a pipe into a reservoir (an exit).

A fluid may flow from a reservoir into a pipe through any number of differently shaped entrance regions as are sketched in Fig. 8.22. Each geometry has an associated loss coefficient. A typical flow pattern for flow entering a pipe through a square-edged entrance is sketched in Fig. 8.23. As was discussed in **Chapter 3**, a vena contracta region may result because the fluid cannot turn a sharp right-angle corner. The flow is said to separate from the sharp corner. The maximum velocity at section (2) is greater than that in the pipe at section (3), and the pressure there is lower. If this high-speed fluid could slow down efficiently, the kinetic energy could be converted into pressure (the Bernoulli effect), and the ideal pressure distribution indicated in Fig. 8.23 would result. The head loss for the entrance would be essentially zero.

Such is not the case. Although a fluid may be accelerated very efficiently, it is very difficult to slow down (decelerate) a fluid efficiently. Thus, the extra kinetic energy of the fluid at section (2) is partially lost because of viscous dissipation, so that the pressure does not return to the ideal value. An entrance head loss (pressure drop) is produced as is indicated in Fig. 8.23. The majority of this loss is due to inertia effects that are eventually dissipated by the shear stresses within the fluid. Only a small portion of the loss is due to the

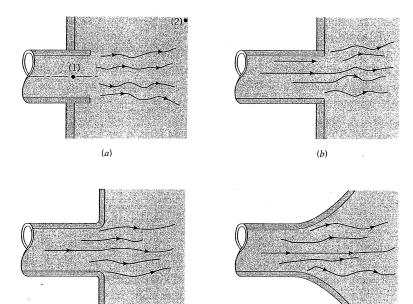
Minor head losses are often a result of the dissipation of kinetic energy.



**B** F I G U R E 8.22 Entrance flow conditions and loss coefficient (Refs. 28, 29). (a) Reentrant,  $K_L = 0.8$ , (b) sharp-edged,  $K_L = 0.5$ , (c) slightly rounded,  $K_L = 0.2$  (see Fig. 8.24), (d) well-rounded,  $K_L = 0.04$  (see Fig. 8.24).



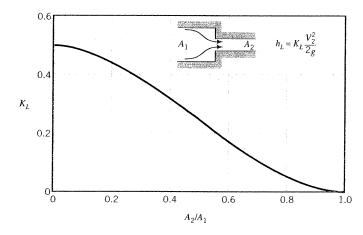
V8.4 Entrance/exit flows



**B** FIGURE 8.25 Exit flow conditions and loss coefficient. (a) Reentrant,  $K_L = 1.0$ , (b) sharp-edged,  $K_L = 1.0$ , (c) slightly rounded,  $K_L = 1.0$ , (d) well-rounded,  $K_L = 1.0$ .

Losses also occur because of a change in pipe diameter as is shown in Figs. 8.26 and 8.27. The sharp-edged entrance and exit flows discussed in the previous paragraphs are limiting cases of this type of flow with either  $A_1/A_2 = \infty$ , or  $A_1/A_2 = 0$ , respectively. The loss coefficient for a sudden contraction,  $K_L = h_L/(V_2^2/2g)$ , is a function of the area ratio,  $A_2/A_1$ , as is shown in Fig. 8.26. The value of  $K_L$  changes gradually from one extreme of a sharp-edged entrance  $(A_2/A_1 = 0)$  with  $K_L = 0.50$  to the other extreme of no area change  $(A_2/A_1 = 1)$  with  $K_L = 0$ .

In many ways, the flow in a sudden expansion is similar to exit flow. As is indicated in Fig. 8.28, the fluid leaves the smaller pipe and initially forms a jet-type structure as it enters the larger pipe. Within a few diameters downstream of the expansion, the jet becomes dispersed across the pipe, and fully developed flow becomes established again. In this process [between sections (2) and (3)] a portion of the kinetic energy of the fluid is dissipated as a result of viscous effects. A square-edged exit is the limiting case with  $A_1/A_2=0$ .



■ FIGURE 8.26 Loss coefficient for a sudden contraction (Ref. 10).

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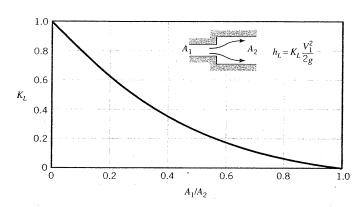


FIGURE 8.27 Loss coefficient for a sudden expansion (Ref. 10).

A sudden expansion is one of the few components (perhaps the only one) for which the loss coefficient can be obtained by means of a simple analysis. To do this we consider the continuity and momentum equations for the control volume shown in Fig. 8.28 and the energy equation applied between (2) and (3). We assume that the flow is uniform at sections (1), (2), and (3) and the pressure is constant across the left-hand side of the control volume  $(p_a = p_b = p_c = p_1)$ . The resulting three governing equations (mass, momentum, and energy) are

8.4

$$A_1V_1 = A_3V_3$$

$$p_1A_3 - p_3A_3 = \rho A_3V_3(V_3 - V_1)$$

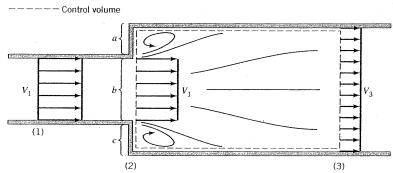
and

$$\frac{p_1}{\gamma} + \frac{V_1^2}{2g} = \frac{p_3}{\gamma} + \frac{V_3^2}{2g} + h_L$$

These can be rearranged to give the loss coefficient,  $K_L = h_L/(V_1^2/2g)$ , as

$$K_L = \left(1 - \frac{A_1}{A_2}\right)^2$$

where we have used the fact that  $A_2 = A_3$ . This result, plotted in Fig. 8.27, is in good agreement with experimental data. As with so many minor loss situations, it is not the viscous effects directly (i.e., the wall shear stress) that cause the loss. Rather, it is the dissipation of kinetic energy (another type of viscous effect) as the fluid decelerates inefficiently.



**EXECUTE** FIGURE 8.28 Control volume used to calculate the loss coefficient for a sudden expansion.

The loss coefficient for a sudden expansion can be theoretically calculated.

The losses may be quite different if the contraction or expansion is gradual. Typical results for a conical diffuser with a given area ratio,  $A_2/A_1$ , are shown in Fig. 8.29. (A diffuser is a device shaped to decelerate a fluid.) Clearly the included angle of the diffuser,  $\theta$ , is a very important parameter. For very small angles, the diffuser is excessively long and most of the head loss is due to the wall shear stress as in fully developed flow. For moderate or large angles, the flow separates from the walls and the losses are due mainly to a dissipation of the kinetic energy of the jet leaving the smaller diameter pipe. In fact, for moderate or large values of  $\theta$  (i.e.,  $\theta > 35^{\circ}$  for the case shown in Fig. 8.29), the conical diffuser is, perhaps unexpectedly, less efficient than a sharp-edged expansion which has  $K_L = (1 - A_1/A_2)^2$ . There is an optimum angle ( $\theta \approx 8^{\circ}$  for the case illustrated) for which the loss coefficient is a minimum. The relatively small value of  $\theta$  for the minimum  $K_L$  results in a long diffuser and is an indication of the fact that it is difficult to efficiently decelerate a fluid.

It must be noted that the conditions indicated in Fig. 8.29 represent typical results only. Flow through a diffuser is very complicated and may be strongly dependent on the area ratio  $A_2/A_1$ , specific details of the geometry, and the Reynolds number. The data are often presented in terms of a pressure recovery coefficient,  $C_p = (p_2 - p_1)/(\rho V_1^2/2)$ , which is the ratio of the static pressure rise across the diffuser to the inlet dynamic pressure. Considerable effort has gone into understanding this important topic (Refs. 11, 12).

Flow in a conical contraction (a nozzle; reverse the flow direction shown in Fig. 8.29) is less complex than that in a conical expansion. Typical loss coefficients based on the downstream (high-speed) velocity can be quite small, ranging from  $K_L = 0.02$  for  $\theta = 30^\circ$ , to  $K_L = 0.07$  for  $\theta = 60^\circ$ , for example. It is relatively easy to accelerate a fluid efficiently.

Bends in pipes produce a greater head loss than if the pipe were straight. The losses are due to the separated region of flow near the inside of the bend (especially if the bend is sharp) and the swirling secondary flow that occurs because of the imbalance of centripetal forces as a result of the curvature of the pipe centerline. These effects and the associated values of  $K_L$  for large Reynolds number flows through a 90° bend are shown in Fig. 8.30. The friction loss due to the axial length of the pipe bend must be calculated and added to that given by the loss coefficient of Fig. 8.30.

For situations in which space is limited, a flow direction change is often accomplished by use of miter bends, as is shown in Fig. 8.31, rather than smooth bends. The considerable losses in such bends can be reduced by the use of carefully designed guide vanes that help direct the flow with less unwanted swirl and disturbances.

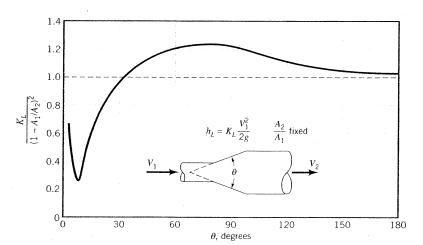
The head loss in a diffuser is strongly dependent on the shape of the diffuser.

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Flow three  $A_2/A_1$ , specified in the effort has

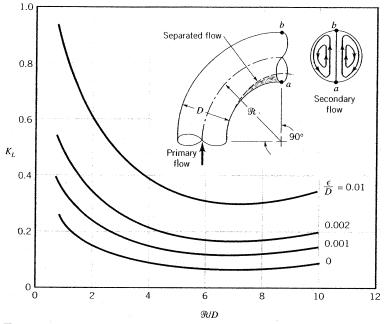


V8.5 Car exhaust system



■ FIGURE 8.29 Loss coefficient for a typical conical diffuser (Ref. 5).

Exte avai coef, dara nent



**FIGURE 8.30** Character of the flow in a 90° bend and the associated loss coefficient (Ref. 5).

Another important category of pipe system components is that of commercially available pipe fittings such as elbows, tees, reducers, valves, and filters. The values of  $K_L$  for such components depend strongly on the shape of the component and only very weakly on the Reynolds number for typical large Re flows. Thus, the loss coefficient for a 90° elbow depends on whether the pipe joints are threaded or flanged but is, within the accuracy of the data, fairly independent of the pipe diameter, flow rate, or fluid properties (the Reynolds number effect). Typical values of  $K_L$  for such components are given in **Table 8.2**. These typical components are designed more for ease of manufacturing and costs than for reduction of the head losses that they produce. The flowrate from a faucet in a typical house is sufficient whether the value of  $K_L$  for an elbow is the typical  $K_L = 1.5$ , or it is reduced to  $K_L = 0.2$  by use of a more expensive long-radius, gradual bend (Fig. 8.30).

Valves control the flowrate by providing a means to adjust the overall system loss coefficient to the desired value. When the valve is closed, the value of  $K_L$  is infinite and no fluid flows. Opening of the valve reduces  $K_L$ , producing the desired flowrate. Typical cross sections of various types of valves are shown in Fig. 8.32. Some valves (such as the

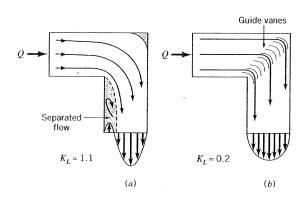


FIGURE 8.31 Character of the flow in a 90° mitered bend and the associated loss coefficient: (a) without guide vanes, (b) with guide vanes.

Extensive tables are available for loss coefficients of standard pipe components.

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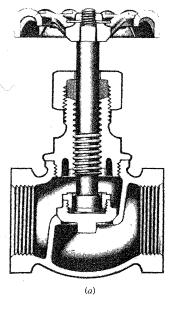
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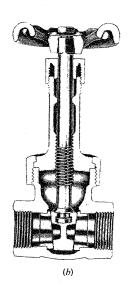
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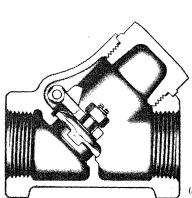
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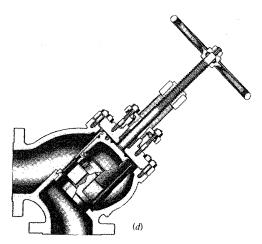
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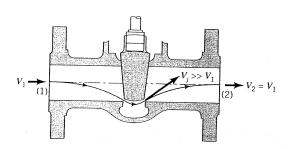








■ FIGURE 8.32 Internal structure of various valves: (a) globe valve, (b) gate valve, (c) swing check valve, (d) stop check valve. (Courtesy of Crane Co., Valve Division.)



■ FIGURE 8.33 Head loss in a valve is due to dissipation of the kinetic energy of the large-velocity fluid near the valve seat.